

# FUEL PROPERTY EFFECTS ON THE UNAIDED COLD STARTING OF A TWO-CYCLE DIESEL ENGINE

INTERIM REPORT BFLRF No. 186

Ву

A.F Montemayor
E.C. Owens
Belvoir Fuels and Lubricants Research Facility (SwRI)
Southwest Research Institute
San Antonio, Texas

**Under Contract to** 

U.S. Army Belvoir Research and Development Center Materials, Fuels and Lubricants Laboratory Fort Belvoir, Virginia

Contract No. DAAK70-85-C-0007

Approved for public release; distribution unlimited

December 1985

DITE FILE CO

85 12 27 045

## Disclaimers

The findings in this report are not to be construed as an official Department of the Army position unless so designated by other authorized documents.

Trade names cited in this report do not constitute an official endorsement or approval of the use of such commercial hardware or software.

# **DTIC Availability Notice**

Qualified requestors may obtain copies of this report from the Defense Technical Information Center, Cameron Station, Alexandria, Virginia 22314.

# Disposition Instructions

Destroy this report when no longer needed. Do not return it to the originator.

	- 1		REPORT DCCUM	ENTATION F	PAGE		
1a. REPORT SEC	URITY CLASSIFIC	CATION		16. FigSTRICTIVE MA	ARKINGS		
Unclassi	fied			None			
2a. SECURITY CL	ASSIFICATION A	UTHORITY			VAILABILITY OF REPO	RT	
N/A				Approved	for public 1	elease;	
		RADING SCHEDULE			tion unlimite		
		REPORT NUMBER(S)		5. MONITORING OR	GANIZATION REPORT	NUMBER(S)	
Interim	Report BF	LRF No. 186					
6a. NAME OF PER			8b. OFFICE SYMBOL (If applicable)	7a. NAME OF MONT	TORING ORGANIZATIO	N	,
		ubricants	(iii appiozoisi				4
	Facility		L	- 1000000000000000000000000000000000000	0 4.710 Co.dol		<u> </u>
6c. ADDRESS (C)	ty, State, and ZIP	Code)		76. ADDRESS (City,	State, and ZIP Codel		
6220 Cu1	ebra Road						
	nio, TX						
			·			CATION NI IA	4060
84. NAME OF FUI		RING	8b. OFFICE SYMBOL		INSTRUMENT IDENTIF		
ORGANIZATIO Relvoir		& Develop-	(If applicable)				<b>&lt;70-85-C-00</b> 07;
ment Cen		a peacioh	STRBE-VF		AK70-82-C-00	01;WD_1	
8c. ADDRESS (C	ity, State, and ZIF	Code)		10. SOURCE OF FU			
m n . 1		22000		PROGRAM	PROJECT	TASK	WORK UNIT
Fort Bel	voir, VA	22000		ELEMENT NO.	NO.	NO.	ACCESSION MO.
, , , , , , , , , , , , , , , , , , , ,				<u></u>	<u> </u>		
12. PERSONAL A	UTHOR(S)		UNAIDED COLD ST	ARTING OF A	TWO-CYCLE D	IESEL E	NGINE (U)
Montemay	or. Alan	F. and Owens	Edwin C.	14 DATE OF BEROR	T (Year, Month, Day)	15 P	AGE COUNT
13a. TYPE OF RE	PORT	13b. TIME CO	n 81 TO Dec 85	December		13. 1	23
Inter1m		FROM LIZ	m ni io nei uz	ресещоет	1703		
16. SUPPLEMEN	TARY NOTATION				and the state of t	. All young in the company of providing the	Make the a regular sear species in the format of a financial contract of
\ /			and the same of th				
17.	COSATIC	ODES	18. SUBJECT TERMS (Co	otinue on reverse if n	ecessary and identify b	v block numbe	<b>Y</b> /
FIELD	GROUP	SUB-GROUP	<del>4</del>	Fuel Pro		, 7 Flash <sup>2</sup>	
FIELD	GROUP	30B-GNOOF	Cetane	Arner Aro	percies;	Diesel	
<b> -/</b>		<u> </u>	Diesel Engine Cold Starting	Viscosit	007;	VCLM D	86 • A
A A DETRACT	/C==#==== == ====	L de la conseque and ide	entify by block numberi	- Tracourte	3		
			iesel 4-53T was	heavily inc	ulated and o	oolad u	hallida a nata
un this	program,	a Decidic D.	nd cooled combus	tion oir mo	a provided	An ext	ornal cranking
COOTAIL	CIICUIAL.	ton system an	ngine at a const	cat vom T	tionti-one to	at fual	e wore blended
motor Wa	ss used Co	curn the el	ngine at a const ng temperature w	ane thme t	for each for	ol IUEL	o were premose.
and a m	inimum una	aided Starti	ng temperature w	as obcarned	tor each fu	er. m	to minimum
regress	ion analys	sis was then	performed in or	der to tera	re ruer brob	( D 05 ~	nd n 2007
starting	g cemperat	ture. Fuel p	properties exami	ned were: Vi	scosicy, ASTM	י יייז ע דירו סט די	100 D 400/
boiling	point ter	mperatures,	cetane number, a	ucoignicion	. cemperature	and r	rasu point.
			% boiling temper				
scatist	ically sig	gnificant imp	pact on minimum	starting te	mperature.	neywo	rds!
						/	· <del></del>
1							
		TY OF ABSTRACT		I -	ECURITY CLASSIFICA	IIUN	
	ASSIFIED/UNLIM		RPT. D DTIC USERS	Unclassi		100 000	OF 6944901
1	RESPONSIBLE IN				(Include Area Code)		CE SYMBOL
	. Schaeke			(703) 66			E-VF
DD FORM 1	473, 84 MAR	1	83 APR edition may be used ur	ntil exhausted.	SECUR	ITY CLASSIF	CATION OF THIS PAGE

#### SUMMARY

The effects of fuel properties on unaided cold startability were evaluated using a Detroit Diesel 4-53T engine. The engine was insulated with approximately 3 inches of fiberglass insulation and a chilled coolant was circulated through its cooling passages. An external cranking motor was used to turn the engine at a constant 150 revolutions per minute (rpm). Cold intake air was provided by using a vortex tube (Hilsch tube cold air generator). Then 21 test fuels were blended and run in the engine. The minimum starting temperature for each fuel was determined by successively cooling the engine and attempting a start at a particular temperature. The minimum starting temperature was the average of two "no-start" temperatures and two "start" temperatures that were no more than 2°C apart. A "no-start" condition was defined as a failure to attain a self-sustaining running state after l minute of cranking at 150 rpm. Analysis of fuel properties and minimum starting temperatures using a statistical analysis program yielded a stable minimum starting temperature prediction equation with cetane number, autoignition temperature, viscosity, and ASTM D 2887 50 percent off temperature as statistically significant independent variables at the 10-percent level of significance. The prediction equation for MUST using the D 2887 boiling temperature is:

> MUST = 32.5445 + 8.6660 \* VISCOSITY -0.1423 \* 50% BOILING POINT -0.6968 \* CETANE + 0.0541 \* AUTOIGNITION TEMP

where Minimum Unaided Starting Temperature (MUST) is in °C, 50 percent boiling point is ASTM D 2887 50 percent off temperature in °C, viscosity is ASTM D 445 kinematic viscosity at 40°C in centistokes, and autoignition temperature is ASTM E 659 in °C. Fuels that experienced fuel delivery (i.e., flow) problems or would not start at room temperature were not included in the analyses.

Accessor For

NTIS CRAPI
DTIC 113
Unann Levi 
Junt 1 2011

By
Diet Euton/
vailability Codes

Vail and/or
Special

## **FOREWORD**

This work was conducted at the Belvoir Fuels and Lubricants Research Facility (SwRI) located at Southwest Research Institute (SwRI), San Antonio, TX, under Contract Nos. DAAK70-82-C-0001 and DAAK70-85-C-0007 during the period January 1981 through December 1984. The work was funded by the U.S. Army Belvoir Research and Development Center, Ft. Belvoir, VA, with Mr. F.W. Schaekel (STRBE-VF) as the contracting officer's representative and Mr. M.E. LePera, Chief of Fuels and Lubricants Division (STRBE-VF), as the project technical monitor.

PROPERTY STRUCTURES VENEZUES

# ACKNOWLEDGMENT

The authors wish to thank Janet Buckingham for conducting the statistical analysis presented herein and to acknowledge the assistance provided by the Belvoir Fuels and Lubricants Research Facility (SwRI) technical and laboratory staff in the performance of the work and the Belvoir F&L Research Facility technical publications group in the preparation of this report.

# TABLE OF CONTENTS

Section	<u>)n</u>	Page
I.	INTRODUCTION	7
II.	BACKGROUND	8
III.	EXPERIMENTAL PROCEDURES AND EQUIPMENT	10
	A. Test Engine and Setup  B. Lubricant  C. Test Fuels  D. Test Procedure	10 13 14 18
IV.	DISCUSSION OF RESULTS	19
٧.	CONCLUSIONS	23
VI.	RECOMMENDATIONS	25
UTT	TOT OF DEPENDENCES	25

# LIST OF ILLUSTRATIONS

1180	16	rage
1 2 3 4	Process for Evaluating Alternative and/or New Fuels  Detroit Diesel 4-53T Test Setup and Engine Specifications  Cooling System Schematic	11 12 13
5	Fuel System Schematic	
6	Starting Data	
7	Observed MUST Versus Cetane Number	
8	Observed MUST Versus Predicted MUST	24
	LIST OF TABLES	
Tab1	<u>e</u>	Page
1	Test Fuel Descriptions	15
2	Test Fuel Properties	16
3	Multiple Linear Regression Statistics for the DDA 4-53T Engine	
	Using ASTM D 2887 Boiling Point Temperatures	22
4	Multiple Linear Regression Statistics for the DDA 4-53T Engine	
•	Using ASTM D 86 Boiling Point Temperatures	22

#### I. INTRODUCTION

Due to the current world oil situation, the U.S. Army wishes to develop a capability to utilize multisource mobility fuels. As the sources of these fuels change, the basic properties of the fuels will also change. The Army currently specifies acceptable property limits for its fuels, but future economic and availability considerations may necessitate expansion of these time-proven limits.

Qualitative fuel property effects on engine startability have long been known and have been incorporated into existing specification limits. Expansion of these limits requires quantitative knowledge of these fuel property effects in order to minimize startability problems with Army vehicles. Figure 1 illustrates a methodology for evaluating new/synthetic fuels to assure that there will be no impairment to overall Army mission. (1)\* The work described in this report falls under the heading of Full-Scale Multicylinder Engine Performance Testing and provides feedback information to the qualification system.

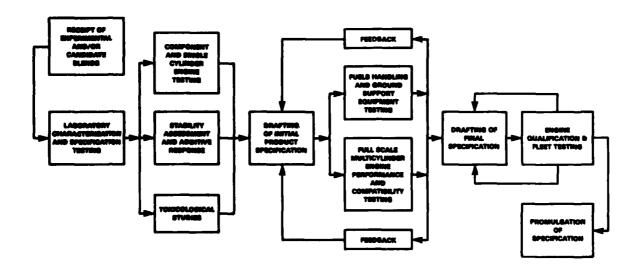


FIGURE 1. PROCESS FOR EVALUATING ALTERNATIVE AND/OR NEW FUELS

<sup>\*</sup> Underscored numbers in parentheses refer to the list of references at the end of this report.

Fuel properties expected to affect low-temperature startability are kinematic viscosity, boiling range, cetane number, autoignition temperature, flash point, and cloud point. Cloud point relates to the ability of the fuel to flow through the screens, hoses, and filters of the fuel system. Viscosity relates to the pumpability of the fuel in both the fuel pump and injection system. Viscosity affects the atomization of the fuel when injected into the combustion chamber. The boiling range of the fuel determines how much of the fuel is vaporized at the temperatures encountered in the combustion chamber under starting conditions. Flash point and autoignition temperature relate to the initiation of combustion in the combustion chamber. Cetane number is a measure of ignition delay and has been determined to play an important role in the starting process. (2-4)

Other parameters affecting startability are engine design, cranking rpm, and ambient temperature. In this study, the cranking rpm was fixed at 150 rpm using a special variable speed cranking system. This speed is used in other studies of low-temperature combustion and is a reasonable minimum low-temperature cranking speed for this engine. (5-7) Ambient temperature was controlled by circulating chilled coolant through the engine, and intake air temperature was controlled with a vortex tube (Hilsch tube).

## II. BACKGROUND

Diesel engines ignite fuel by spraying the fuel into high-temperature, high-pressure air and allowing the fuel to autoignite. The high temperatures are generated by isentropic compression of the intake air charge. A variety of factors affects the temperature of the air at the time of fuel injection. These factors include ambient temperature, effective engine compression ratio, cranking speed, duration of cranking time, and injection timing. Properties of the fuel have a negligible effect on these variables.

However, fuel properties do play an important role in the autoignition process. As the fuel is injected into the heated air in the combustion

chamber, the fuel is atomized into droplets of varying sizes and dispersed as plumes of droplets and entrained air. The droplets are then heated by the air and partially evaporated. Also during this period, chemical reactions are occurring, which eventually lead to exothermic oxidation reactions between the fuel components and oxygen. If these reactions liberate sufficient heat quickly enough, the increased temperature further accelerates these combustion processes, and ignition is said to have occurred. Obviously, this ignition process is complex and is not understood in its entirety. However, the total ignition delay period from beginning of injection to ignition (injection, atomization, evaporation, mixing) consists of a physical delay period, and a chemical delay period (chain breaking, radical generation, oxidation) before ignition. This total process is currently quantitized by cetane number.

The cetane number of a fuel is determined by operating a special test engine on the fuel and measuring the engine compression ratio necessary to produce a 13-degree ignition delay at carefully controlled operating conditions. (8,9) Several researchers have shown that the cetane number is a fuel characteristic that correlates strongly with the ease of starting of diesel engines. (2,4)

MANAGER ACCOUNTS AND SOLVE

These include viscosity effects on atomization, boiling point distribution which influences evaporization, chemical composition, etc. It has been realized that the peculiarities of the CFR cetane engine may mean that other diesel engines may not respond identically to fuel characteristic variations. Nevertheless, cetane number has been an acceptable indicator of fuel ignition quality for field applications. This was particularly true when diesel fuels were generally uniform in physical properties and the cetane number specification was maintained at a sufficiently high level to provide considerable starting margin under most conditions.

However, with the recent interest in nonhydrocarbon diesel fuels, and non-petroleum hydrocarbons, indications have arisen that the cetane number may not be a totally adequate measure of rapidity of autoignition. This difficulty is further aggravated by a continuing decline in crude oil quality,

making maintenance of high cetane numbers increasingly expensive. For the military, the prospect of operating its diesel equipment on fuels that fail to meet current procurement specifications has placed additional emphasis on developing more detailed knowledge of the engine and fuel factors influencing engine startability.

One objective of this project was to more closely determine the actual fuel characteristics required to ensure startability of the Detroit Diesel 4-53T engine at low temperatures. A second objective was to determine if the design features of this engine—injection system, air motion, heat transfer rates, etc.—have changed the relative importance of fuel properties in determining the total ignition delay. Such changes in the response of this engine, relative to that of the CFR cetane engine, would result in cetane number being an inadequate predictor of minimum starting temperatures.

# III. EXPERIMENTAL PROCEDURES AND EQUIPMENT

# A. Test Engine and Setup

THE REPORT OF THE PARTY OF THE

The Detroit Diesel 4-53T engine was chosen as the test engine due to its relatively high recommended minimum unaided starting temperature, low fuel consumption, small size, and density of the 53 series engines in the Army's fleet. Detroit Diesel recommends the use of starting aids at temperatures less than 4°C (40°F).(10) The engine and a list of specifications are shown in Figure 2 (note that the picture is not from this test program). The engine was insulated with 3 inches of fiberglass batting, and plumbing was installed to circulate chilled coolant through the engine. The circulating coolant was 60 percent by weight ethylene glycol and 40 percent by weight water. The circulating coolant was chilled by passing through a coil immersed in Stoddard solvent and dry ice. Temperatures in the engine were controlled by varying the flow rates of the circulating coolant and controlling the amount of dry ice in the Stoddard solvent tank. Figure 3 depicts the cooling system.



Model: 4-53T (5047-5340)

Engine Type: Two-cycle compression ignition, direct injection, uniflo scavenging, turbosupercharged

Cylinders: 4, inline

Displacement: 3.48 L (212 in. 3)

Bore: 9.84 cm (3.87 in.) Stroke: 11.43 cm (4.5 in.) Compression Ratio: 18.7:1

Fuel Injection: DD 5A60 unit injectors

Rated Power: 127 kW (170 BHP)

at 2500 rpm

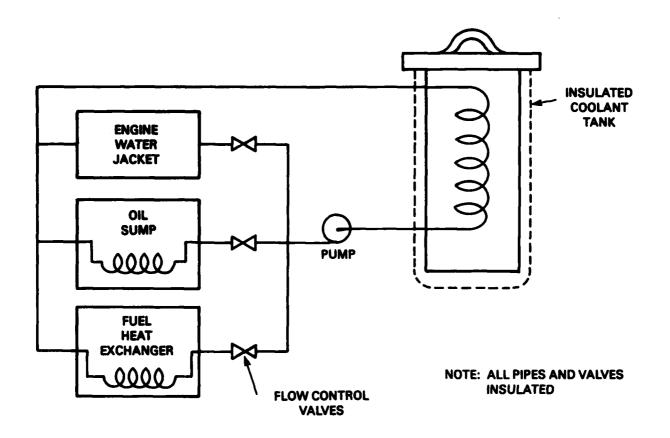
SAMOON SALAN

CONTRACTOR STREET, STR

Rated Torque: 545 N·M (402 lb-ft)

at 1800 rpm

FIGURE 2. DETROIT DIESEL 4-53T TEST SETUP
AND ENGINE SPECIFICATIONS



Marines on the same standards

ANNESS SERVICE PROPERTY BROKES CONTROL

FIGURE 3. COOLING SYSTEM SCHEMATIC

Inlet air temperature was controlled by running dried compressed air through a vortex tube (Hilsch tube). Figure 4 depicts the inlet air system. In practice, cold air was allowed to escape from the inlet air pipe until the engine was cranked. More air was supplied to the inlet air tube than the engine would take in. Excess air vented through the inlet air tube and was observed with a telltale (see Figure 3). No air filter was used for this test.

Engine cranking speed was controlled at 150 rpm with a variable speed external cranking system. The cranking system was equipped with an over-running clutch (sprag clutch) so that the engine could start and come up to idle speed.

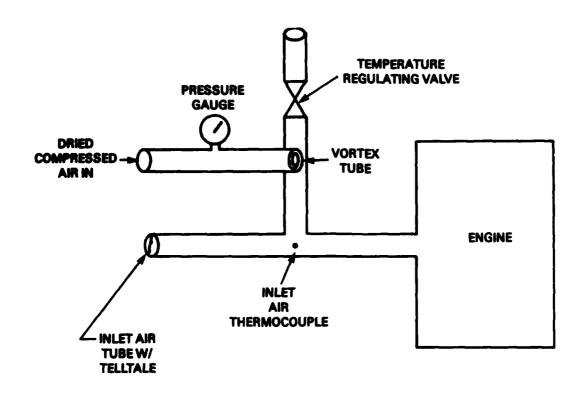


FIGURE 4. INLET AIR SYSTEM SCHEMATIC

The fuel system consisted of the normal engine-mounted lines, filters, and pumps except that the primary fuel filter was removed. Fuel supply was from a 1-gallon reservoir resting on a fuel scale at engine height. Return fuel was routed back to the 1-gallon container. Fuel lines were supported using a laboratory stand such that the fuel weight in the reservoir could be determined before and after the start attempts. The fuel filter and fuel heat exchanger were cooled in order to assure that the test fuel would be at or near the test temperatures. The fuel supply can was not cooled due to the problems of accurately weighing an insulated cooled container. The fuel system schematic is illustrated in Figure 5.

## B. Lubricant

SPECIAL NUMBER

ANY ACCRETO SECONDS (SECONDS (OCCUPANT) AND

The engine lubricant used throughout the test was a qualified MIL-L-46167 Arctic engine oil. This lubricant was chosen as being representative of

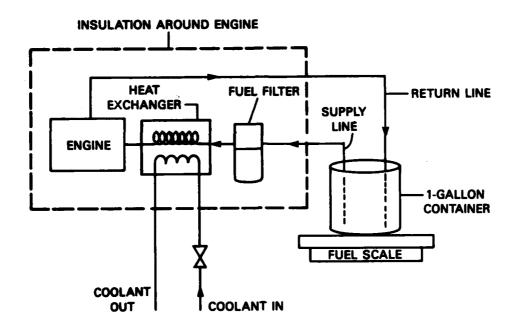


FIGURE 5. FUEL SYSTEM SCHEMATIC

Army field use and providing good low-temperature viscosity and flow characteristics.

## C. Test Fuels

TARLEGIE COLLEGER MENTANDO SASSASSE CACCACAC

Twenty-one hydrocarbon fuels were evaluated in this program. Ten of the fuels were designated neat fuels, meaning that they contained no additional additives nor were blends of different neat fuels. Eleven of the fuels were blends of the neat fuels and several additives.

Summaries of the test fuel descriptions and test fuel properties are shown in Tables 1 and 2, respectively. Neat fuels were Caterpillar 1-H/1-G reference fuel, Jet A, BTX Bottoms\*, VV-F-800 grade DF-1, Stoddard solvent, kerosene, MIL-T-5624 grade JP-4, Type 1 referee fuel, Type 2 referee fuel, and gas oil. The Type 1 and Type 2 referee fuels listed in Table 1 are

<sup>\*</sup> BTX bottoms consist of  $C_Q$  and higher aromatic compounds and are the end product of a petroleum refining process in which benzene, toluene, and xylene are extracted.

# TABLE 1. TEST FUEL DESCRIPTIONS

Test Fuel Code	AL Number	Description
1	11372	Caterpillar 1-H/1-G reference fuel
2	10582	Jet A
3	10716	BTX Bottoms ( $C_{q}$ and heavier aromatics)
4	9294	VV-F-800 grade DF-1
5	11232	Stoddard solvent
6	11233	Kerosene
7	10583	MIL-T-5624 grade JP-4
8	10849	Gas oil
9	11514	JP-4 + 15 percent PAO
10	11515	Stoddard solvent + 15 percent PAO
11	11516	Jet A + 15 percent PAO
12	11517	DF-1 + 15 percent PAO
13	10999	Type 1 referee fuel
14	11017	Type 2 referee fuel
15	11571	JP-4 + 0.11 percent Amyl nitrate
16	11636	Type 1 referee fuel + 1.1 percent amyl nitrate
17	11637	Type 2 referee fuel + 0.19 percent amyl nitrate
18	11638	Jet A + 3.0 percent BTX bottoms
19	11639	Caterpillar 1-H/1-G + 35.0 percent BTX bottoms
20	11640	DF-1 + 23.0 percent BTX bottoms
21	11641	Gas oil + 22.0 percent BTX bottoms

TEST FUEL PROPERTIES TABLE 2.

CONTROL CONTROL CONTROL CONTROL

Determination	Method					Test	Pue!				
		11372	10582	91201	7626	11232 11	11233	10583	10649	71217	11515
Kinematic Viscosity at	•	1	,	;	;	;		;	•	,	,
40°C, cSt Distillation Tempera-	D 445	3.0	1.5	0.75	1.95	1.07	1.56	0.78	7.91	1.12	1.39
ture @ WtZ Off, "C	D 2867					,		,	,	,	
186		125.7	140.6	9.66	184.6	85.5	102.2	21.1	215	*	124.8
10		220.5	179.5	158.2	207.8	105.7	173.7	75.3		85.1	151.9
20		239.9	193.4	162.6	215.7	167.4	206.5	93.3	<b>%</b>	101.5	163.3
2		256.9	201.3	165.6	221.8	175.2	219.8	115.8	323.3	125.6	169.8
9		270.8	210	168.5	229.3	182.5	231	136.2	337.2	161.2	174.1
2		282.8	218	170.1	236.9	191.2	236.2	162.6	350.7	191.2	177.8
09		295.7	226.3	175.1	245.2	197.6	239.2	188.3	364.7	214	183.7
2		307.2	234.6	180.9	257.2	204.5	244.4	209.3	379.5	235.2	193.4
28		321.8	244.4	185.3	274.8	214	253.3	226.3	397.9	270.1	207.2
06		343	257.2	189.7	304.8	221.3	258.4	246.7	424.3	481.2	474
ä		405.5	312	218.8	362.5	248.8	284.2	309.1	496.7	572.1	566.5
Distillation Tempera-											
ture @ WtZ Off, "C	D 86*										
187		196.9	178.1	165.5	209.9	142.2	161.4	75.8	265.5	82.5	156.6
22		226.9	190.4	170.9	217.3	163.4	184.8	87.4	288.5	96.0	164.5
101		237.8	196.3	172.3	221.5	169.4	197.3	97.0	301.1	105.9	168.7
20%		251.6	202.7	172.2	225.0	176.6	214.1	110.7	316.6	119.9	173.0
30%		261.9	207.6	171.2	228.0	101.8	224.3	126.1	327.9	139.6	175.0
202		278.2	215.0	168.8	233.4	188.8	232.8	160.9	344.5	188.8	175.8
70%		295.2	225.1	172.9	250.5	196.7	234.9	198.6	365.5	231.0	188.9
202		304.3	230.0	172.3	264.9	202.7	238.2	214.0	378.1	288.3	238.7
206		322.5	241.8	175.5	290.1	207.9	241.0	232.5	400.9	452.5	445.3
202		351.9	272.5	193.7	328.8	231.1	257.6	267.6	434.6	484.5	477.1
Cetane Number	D 613	2	45	~	57	1	47	35	19	43	64
Autoignition Tempera-											
ture, "C	E 659	245	250	475	245	255	265	245	235	161	185
Cloud Point, *C		ģ	9	4	-21	9	- 38 - 38	0 <del>9-</del> >	74	-54	9-
Flash Point, °C	D 93	77.0	63.6	43.6	82.6	36.0	25	-24.0	135.6	-21.0	43.0
Annians Unaided Start- ing Temperature, *C**	AFLRL	0.4	-1.0	****	9.6	2.8	4.6-	1.0	-9.0###	4.2	-7.6
•											

\* Predicted values using ASTM D 2837 data and a correlation from ASTM STP 577 (13)
\*\* Starting temperatures are the average of two start alrbox temperatures and two no-start alrbox temperatures that are no more than 2°C

TEST FUEL PROPERTIES (CONT'D) TABLE 2.

		11516	11517	10999	11017	11571	11636	11637	11638	11639	11640	11641
Kinematic Viscosity at 40°C, cSt Distillation Tempera-	D 445	2.07	2.57	0.76	3.74	0.82	0.78	3.73	1.46	1.80	1.49	3.76
ture e WtZ Off, "C IBP 10	D 2887	143.1	189.7	25.7	114.7	30.8	28.3	116	138.1	135.1	147.6	150.6
30 02		198.4	217.6	106.2	267.9	101.5	110.5	269.9	188.7	170.1	179.7 203.8	187:2
\$ <b>\$</b> \$		215.4	232.1	160.5	294 306.1	143.1	160	295.5	205.2	231.7	212.6	311.4
09		234.4	252	178.5	318	194.9	175.2	328.5	219.4	271.4	227.8	346.6
2 22 5		265.6	317.2	214.7	344.6	231.4	212.1	341.2	237.1	306.9	251.7	378.7
RP Tenneral		565.8	567.2	300.8	441.6	308.3	297.8	426.3	298.6	392.9	351.8	438.3
ture & Wtz Off. 'C	D 86*											
201		182.3	213.8	83.1	208.7	6.78	83.5	209.9	173.9	165.7	169.8	149.8
77 10 <b>%</b>		201.5	224.3	107.5	264.2	105.9	108.4	265.7	191.0	177.0	181.2	165.2
202		208.2	226.9	125.2	277.5	118.4	127.9	279.0	198.4	186.6 203.6	196.6	238.1
202		220.6	234.8	164.4	301.0	168.0	162.6	301.1	209.9	250.1	215.6	324.4
70% 80%		240.5	322.4	189.4 203.9	317.4	203.5 218.9	187.6 201.3	315.0 320.9	219.0 223.7	274.4	242.7	345.8 361.0
<b>206</b>		442.6	444.2	224.0	339.5	235.9	221.3	335.6	235.7	310.9	269.3	377.2
Cetane Number	D 613	84	3	82	35	36	<b>2</b>	37	<b>4</b> 2	8	*	9
Autoignition Tempera- ture. *C	E 659	185	179	202	210	190	190	204	185	061	183	<b>08</b> 1
Cloud Point, °C	D 2500	-48	-16	<b>9</b>		9	<del>د</del> ک	0	9	-12	<b>77</b>	9
Flash Point, "C	D 93	0.49	<b>%</b>	-21.0	32.0	-22.0	-2.0	37.0	55.0	20.0	60.5	60.5
ing Temperature, "Cat	APLAL	-6.0	-9.8	9.0	12.0	0.7	4.0	3.5	4.6-	ř	4.5	-12.5***

<sup>\*</sup> Predicted values using ASIM D 2837 data and a correlation from ASIM STP 577 (13)

proposed Army referee fuels. Type 1 referee fuel was blended to exhibit high volatility and low cetane number. Type 2 referee fuel was blended for low volatility, low cetane number, and high viscosity. A polyalphaolefin (PAO) compound (6 cSt at 210°F) was added to four of the neat fuels to increase the viscosity of the fuels. (11) The PAO compound also increased the cetane number of the fuels. Amyl nitrate was added to three of the neat fuels in order to increase their cetane numbers. (12) BTX bottoms were added to four of the neat fuels in order to decrease their cetane numbers.

Viscosity range for the test fuels was 0.78 to 7.91 cSt at 40°C. Ten percent boiling point by D 2887 ranged from 84.1° to 285°C. Cetane numbers for the fuels were 5.0 to 61.0. Autoignition temperatures by E 659 were from 180° to 475°C. Cloud points by D 2500 were from <-60°C up to 14°C. Flash points by D 93 ranged from -22° to 135.6°C. Blending of the additives was done on a volume basis. Test fuels were selected to obtain wide variations in the selected fuel properties.

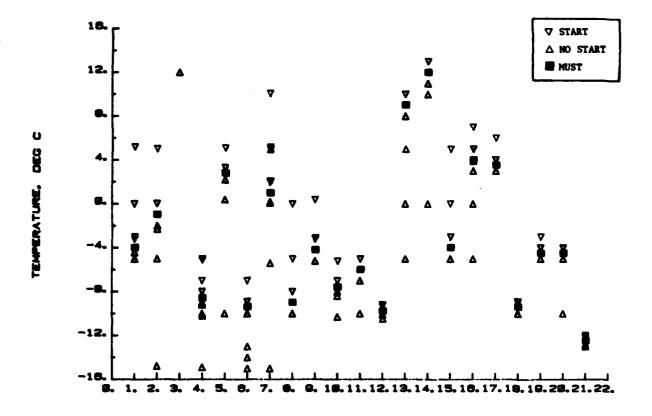
## D. Test Procedure

Dry ice was added to the insulated coolant tank in order to begin the cooling process. The coolant pump was turned on (see Figure 3), circulating the chilled coolant through the engine block, oil sump heat exchanger, and fuel heat exchanger. Temperatures at the airbox, oil sump, and fuel heat exchanger were adjusted to the desired test temperature by adding dry ice to the coolant tank and controlling the flow of coolant with individual flow control valves. Temperature stabilization required approximately 3 hours of dry ice and flow adjustments. After the desired test temperatures were achieved, the vortex tube on the inlet air stream was pressurized and adjusted to the test temperature. A sensor on the inlet air pipe assured that sufficient quantities of cold combustion air were supplied to the engine. Pretest temperatures at the airbox, exhaust manifold, oil sump, water jacket, fuel heat exchanger, and inlet air pipe were recorded. In addition, the weight of fuel in the fuel reservoir was recorded.

The engine was cranked at 150 rpm in the full rack position for a maximum of l minute. A finite cranking time was selected to improve reproducibility. With the external cranking system employed, the engine could be motored for long periods and thus reduce the minimum starting temperature. This is because extended cranking would heat the combustion chamber through ring/ liner friction and heat transfer from the air charge. This heating would reduce the heat lost from the air during the compression event, raising the air temperature at injection and improving the starting. However, in the field the cranking time is limited by the available battery energy and the time to overheat the cranking motor. The 1-minute continous cranking period was chosen as a compromise of these factors. If the engine started and continued running after 1 minute (or less) of cranking, then this was considered a "start" and the after test information recorded. If the engine failed to start or continue running after I minute of cranking, then this was considered a "no-start" condition, and the after test information was This procedure was repeated at different target temperatures until two "start" and two "no-start" runs were completed that were no more than 2°C apart. The average of the two start and two no-start airbox temperatures was considered to be the minimum unaided starting temperature (MUST) for that fuel. Fuel flushing of the injection system consisted of running I gallon of the next test fuel through the engine with the fuel return line routed to a dump can. This was done at room temperature with the engine running under its own power. In all, 21 test fuels were run using this procedure.

# IV. DISCUSSION OF RESULTS

Figure 6 graphically depicts the results of the cold starting tests. Three types of results were obtained using this procedure. The first type is demonstrated by test fuel 3(AL-10716) (see Tables 1 and 2) which had an extremely low cetane number of 5. This fuel failed to start in the engine at room temperature (12°C). Failure to start was due to the high-aromatic, low-cetane, high-autoignition nature of the fuel. This fuel was excluded from statistical analysis since no numerical value could be placed on its startability.



TEST FUEL CODE

FIGURE 6. STARTING DATA

The second type of result manifested itself in test fuels 8(AL-10849) and 21(AL-11641). These fuels failed to start because of flow-related problems. Starting with these fuels was limited by the cloud points of the fuels. This flow failure was detected by monitoring the before-test fuel reservoir weight and the after-test fuel reservoir weight. No weight change on these fuels indicated that no fuel was consumed (injected) during the start attempt. Since this study was to investigate combustion effects, not pumpability effects, these fuels were not included in the statistical analysis.

The third type of result obtained was the true combustion-related MUST. Eighteen of the test fuels exhibited valid MUST's. MUST's for all the test fuels are shown with fuel properties in Table 2. The SAS (14) stepwise linear regression computer software package was utilized to find the best set of variables to predict MUST. The set of independent variables included viscosity, cetane number, autoignition temperature, flash point, and the boiling point temperature distribution (IBP, 10, 20, 30, 40, 50, 60, 70, 80, 90, EP). Correlations were examined among variables entering the prediction equation through the stepwise regression program. Those variables with high correlation were analyzed, and the appropriate variable was dropped from the set of independent variables. Scatterplots were analyzed for possible variable transformations, and none were found to be necessary.

ASTM D 2887 and ASTM D 86 boiling point temperatures were used in separate stepwise linear regression analyses. There did not appear to be any advantage of using one method over the other in predicting MUST for this set of data. Although cloud point was determined for each fuel, it was not included in the analysis because of the nou-numeric nature of some of the results (e.g., the less than -60°C results). The prediction equation for MUST using the D 2887 boiling point temperature is:

The prediction equation for MUST using the D 86 boiling point temperature is:

In the equations, MUST is in °C, 50 percent Boiling Point is ASTM D 2887 or ASTM D 86 50 percent off temperature in °C, viscosity is ASTM D 445 kinematic viscosity at 40°C in cSt, autoignition temperature is ASTM E 659 in °C, and cetane is ASTM D 613 cetane number.

Tables 3 and 4 summarize the statistics associated with predicting MUST using the D 2887 and D 86 boiling point temperatures, respectively.

TABLE 3. MULTIPLE LINEAR REGRESSION STATISTICS FOR THE DDA 4-53T ENGINE USING ASTM D 2887 BOILING POINT TEMPERATURES

Engine: DDA 4-53T Data Points: 18

Multiple R-Square: 0.7811

Standard Error of Estimate: 3.4376

Variable	Coefficient	Standard Error	T	P*
Intercept	32.5445	-	-	-
Viscosity	8.6660	3.0175	2.8720	0.0131
50% BP	-0.1423	0.0621	-2.2920	0.0392
Cetane Number	-0.6968	0.1088	-6.4050	0.0001
Autoignition Temperature	0.0541	0.0298	1.8130	0.0929

<sup>\*</sup> The P value represents the probability that a T statistic would obtain a greater absolute value than the observed given that the true parameter (coefficient) is zero. The T statistic is a method for expressing the significance of a coefficient by its estimated standard error. A P value of 0.10 represents a 10 percent level of significance.

TABLE 4. MULTIPLE LINEAR REGRESSION STATISTICS FOR THE DDA 4-53T ENGINE USING ASTM D 86 BOILING POINT TEMPERATURES

Engine: DDA 4-53T Data Points: 18

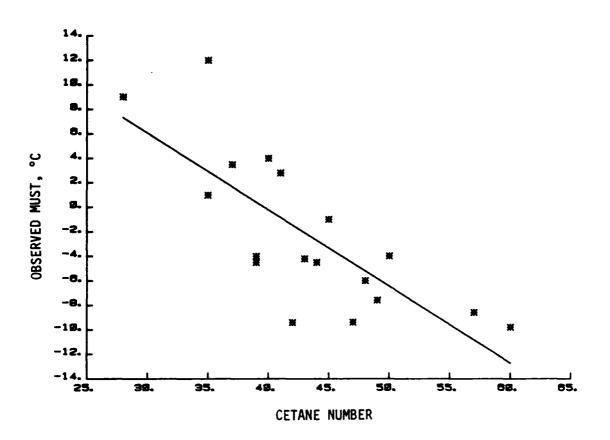
Multiple R-Square: 0.7812

Standard Error of Estimate: 3.4363

Variable	Coefficient	Standard Error	T	P*
Intercept	32.8953	-	-	-
Viscosit <del>y</del>	8.6748	3.0176	2.8750	0.0130
50% BP	-0.1459	0.0636	-2.2950	0.0390
Cetane Number	-0.6968	0.1088	-6.4080	0.0001
Autoignition Temperature	0.0541	0.0298	1.8140	<b>0.0928</b>

<sup>\*</sup> The P value represents the probability that a T statistic would obtain a greater absolute value than the observed given that the true parameter (coefficient) is zero. The T statistic is a method for expressing the significance of a coefficient by its estimated standard error. A P value of 0.10 represents a 10 percent level of significance.

Figure 7 is a plot of observed MUST against cetane number for the eighteen test fuels. Obviously, something more than cetane number is affecting the MUST. A regression analysis using cetane number alone to predict MUST yielded an R<sup>2</sup> fit of only 0.599. Using all the fuel properties contained in Equation 1 yields an R<sup>2</sup> fit of 0.7811 as shown in Table 3. Figure 8 plots observed versus predicted values of MUST using Equation 1. The line in Figure 8 represents the predicted equals observed case.



THE PERSON NAMED AND PARTY AND PARTY AND PARTY.

a respected respective rec

FIGURE 7. OBSERVED MUST VERSUS CETANE NUMBER

# V. CONCLUSIONS

It is possible to quantify the unaided cold-starting characteristics of diesel engines using fuel properties as the independent variables.

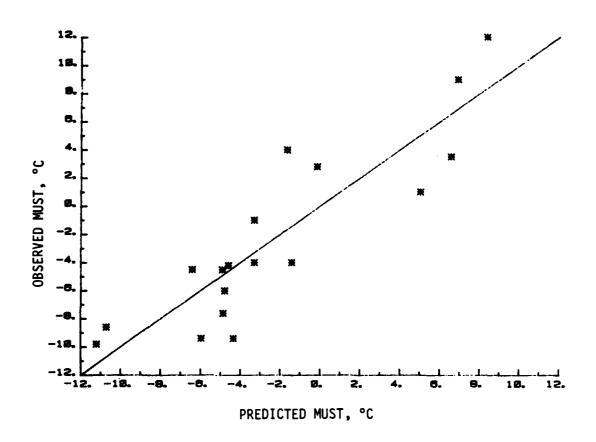


FIGURE 8. OBSERVED MUST VERSUS PREDICTED MUST

- Viscosity, volatility, autoignition temperature, and cetane number are statistically significant (P <0.10) predictors of minimum unaided starting times (MUST's).
- Charts, equations, or computer programs for predicting the MUST's of fuels could be useful for the utilization of off-specification, alternate, or captured fuels in cold climates.

#### VI. RECOMMENDATIONS

- Future MUST tests should utilize a refrigerated box to obtain cold ambient temperatures. This would afford more uniform, repeatable, and realistic conditions.
- Additional MUST tests should utilize more fuels (to increase the statistical sample size), examine more fuel properties, and utilize different types of diesel engines.

# VII. LIST OF REFERENCES

- 1. LePera, M.E., "The U.S. Army's Alternative and Synthetic Fuels Program," Army Research, Development, and Acquisition Magazine, 18-20, September-October 1980.
- Needham, J.R., "The Influence of Fuel Variables on the Operation of Automotive Open and Pre-Chamber Diesel and Spark-Ignited Stratified Charge Engines: A Literature Study Covering Petroleum and Syncrude-Derived Fuels," U.S. Department of Energy Report No. DOEICE/500Z1-1, September 1980.
- 3. Urban, C.M. and Gray, J.T., "A Study of Marginal Compression Ignition," presented at Society of Automotive Engineers Combined Fuels and Lubricants, Powerplant and Transporation Meetings, Pittsburgh, PA, 30 October-3 November 1967.

STATE STATE OF THE PARTY OF THE

- 4. Johnston, A.A., Moffitt, J.V., and Gray, J.T., "Fuel-Engine Sensitivity Studies in the LD/LDS-465 Engine," Summary Report AFLRL No. 58, February 1975.
- 5. Mayer, W.E. and DeCarolis, J.J., "Compression Temperatures in Diesel Engines Under Starting Conditions," Society of Automotive Engineers Transactions, Vol. 70, pp. 163-174, 1962.

- 6. Taniguchi, B.Y. and Benson, J.D., "Cold Weather Fuel Requirements of Oldsmobile Diesels," Society of Automotive Engineers Paper No. 800223, Detroit, MI, 25-29 February 1980.
- 7. Springer, K.J., "An Investigation of the Combustion Process in a Compression Ignition Engine During Low-Speed Starting Conditions," M.S. Thesis for Trinity University, May 1966.
- 8. Boldt, K., and Hall, B.R. (eds), "Significance of Tests for Petroleum Products," ASTM Special Technical Publication 7C, American Society for Testing and Materials, 1916 Race Street, Philadelphia, PA, pp. 96-97.
- 9. 1984 Annual Book of ASTM Standards, American Society for Testing and Materials, 1916 Race Street, Philadelphia, PA, Vol. 5.04, Section II, pp. 235-305, 1985.
- 10. Series 53 Service Manual for Detroit Diesel Engines, No. 6SE201, prepared by Detroit Diesel Allison Division, Detroit, MI, April 1981.
- 11. Technical Data Pamphlet No. 20056-A, prepared by Gulf Oil Chemicals Company, Houston, TX, 31 January 1977.
- 12. Olson, D.R., Meckel, N.T., and Quillian, Jr., R.D., "Combustion Characteristics of Compression Ignition Engine Fuel Components," Society of Automotive Engineers Paper No. 263A, Tulsa, OK, 2-4 November 1960.
- 13. "Calculation of Physical Properties of Petroleum Products From Gas Chromatographic Analyses," ASTM Special Technical Publication 577, American Society for Testing and Materials, 1916 Race Street, Philadelphia, PA, 1974.
- 14. SAS, SAS Institute Inc., Cary, NC.

THE PARTY OF THE PROPERTY OF THE PARTY OF TH

# DISTRIBUTION LIST

DEPARTMENT OF DEFENSE		DIRECTOR US ARMY MATERIEL SYSTEMS
DEPT. OF DEFENSE ATTN: DASD, LMM (MR DYCKMAN)	1	ANALYSIS ACTIVITY ATTN: AMXSY-CM (MR NIEMEYER)
WASHINGTON DC 20301		ABERDEEN PROVING GROUND MD 21005
CDR DEFENSE FUEL SUPPLY CTR		CDR US READINESS COMMAND
ATTN: DFSC-T (MR. MARTIN) CAMERON STATION	1	ATTN: J4-E MACDILL AIR FORCE BASE FL 33608
ALEXANDRIA VA 22314		
DEPT. OF DEFENSE	•	AMC MATERIEL SUPPORT ACTIVITY
ATTN: DUSDRE (RAT), (Dr. Dix) ROOM 3-D-1089, PENTAGON	1	ATTN: AMXTB-T (MR STOLARICK)  FORT LEWIS WA 98433
WASHINGTON DC 20301		HQ, 172D INFANTRY BRIGADE (ALASKA)
DEPARTMENT OF THE ARMY		ATTN: AFZT-DI-L 1 AFZT-DI-M 1
HG, DEPT OF ARMY		DIRECTORATE OF INDUSTRIAL OPERATIONS
ATTN: DALO-TSE (COL NAJERA) DALO-AV	1 1	FORT RICHARDSON AK 99505
DAMA-ART (LTC RINEHART) DAMA-ARZ-E (DR CHURCH)	1 1	CDR US ARMY GENERAL MATERIAL &
WASHINGTON DC 20310		PETROLEUM ACTIVITY ATTN: STRGP-G (COL CLIFTON)  1
CDR		NEW CUMBERLAND ARMY DEPOT
U.S. ARMY BELVOIR RESEARCH AND DEVELOPMENT CENTER		NEW CUMBERLAND PA 17070
	10	CDR
STRBE-WC	2	US ARMY COLD REGION TEST CENTER
FORT BELVOIR VA 22060		ATTN: STECR-TA 1 APO SEATTLE 98733
CDR US ARMY MATERIEL DEVEL &		CDD
READINESS COMMAND	•	US ARMY RES & STDZN GROUP
ATTN: AMCLD (DR ODOM) AMCDE-SG	1	(EUROPE) ATTN: AMXSN-UK-RA 1
AMCDE-SS	I	BOX 65
5001 EISENHOWER AVE ALEXANDRIA VA 22333		FPO NEW YORK 09510
CDR		CDR US ARMY FORCES COMMAND
US ARMY TANK-AUTOMOTIVE CMD		ATTN: AFLG-REG
ATTN: AMSTA-RG (MR WHEELOCK)	3	AFLG-POP 1
AMSTA-RC	1	FORT MCPHERSON GA 30330
AMSTA-MT	1	
AMSTA-MLF (MR KELLER)	2	CDR
AMSTA-GBP (MR MCCARTNEY) WARREN MI 48090	2	US CENTRAL COMMAND
WARREN MI 40070		ATTN: CINCCEN/CC J4-L 1 MACDILL AIR FORCE BASE FL 33608

1/85 AFLRL No. 186 Page 1 of 4

CDR US ARMY YUMA PROVING GROUND ATTN: STEYP-MLS-M (MR DOEBBLER) YUMA AZ 85364	1	CDR US ARMY LEA ATTN: DALO-LEP 1 NEW CUMBERLAND ARMY DEPOT NEW CUMBERLAND PA 17070
PROGRAM MANAGER, BRADELY FIGHTING VEHICLE SYS ATTN: AMCPM-FVS-M WARREN MI 48090	1	HQ, EUROPEAN COMMAND ATTN: J4/7-LJPO (LTC LETTERIE) VAIHINGEN, GE APO NY 09128
PROD MGR, M113 FAMILY OF VEHICLE ATTN: AMCPM-M113-T WARREN MI 48090 PROJ MGR, MOBILE ELECTRIC POWER ATTN: AMCPM-MEP-TM 7500 BACKLICK ROAD	1	CDR US ARMY GENERAL MATERIAL & PETROLEUM ACTIVITY ATTN: STRGP-PW (MR PRICE) BLDG 247, DEFENSE DEPOT TRACY TRACY CA 95376
PROJ OFF, AMPHIBIOUS AND WATER CRAFT		PROJ MGR, LIGHT ARMORED VEHICLES ATTN: AMCPM-LA-E 1 WARREN MI 48090
ATTN: AMCOP-AWC-R 4300 GOODFELLOW BLVD ST LOUIS MO 63120	1	HQ, US ARMY T&E COMMAND ATTN: AMSTE-TO-O 1 ABERDEEN PROVING GROUND MD 21005
CDR US ARMY EUROPE & SEVENTH ARMY ATTN: AEAGG-FMD AEAGD-TE APO NY 09403	1	CDR, US ARMY TROOP SUPPORT COMMAND AMCPM-PWS (LTC FOSTER) 4300 GOODFELLOW BLVD ST LOUIS MO 63120
CDR THEATER ARMY MATERIAL MGMT CENTER (200TH) - DPGM DIRECTORATE FOR PETROL MGMT ATTN: AEAGD-MMC-PT-Q APO NY 09052	1	TRADOC LIAISON OFFICE ATTN: ATFE-LO-AV 1 4300 GOODFELLOW BLVD ST LOUIS MO 63120
CDR US ARMY RESEARCH OFC AMXRO-EG (DR MANN) P O BOX 12211	1	HQ US ARMY TRAINING & DOCTRINE CMD ATTN: ATCD-SL-5 (MAJ JONES)  FORT MONROE VA 23651
PROG MGR, TACTICAL VEHICLE ATTN: AMCPM-TV WARREN MI 48090	1	CDR US ARMY TRANSPORTATION SCHOOL ATTN: ATSP-CD-MS (MR HARNET) 1 FORT EUSTIS VA 23604
WITH TOUTO		PROJ MGR, PATRIOT PROJ OFFICE US ARMY MATERIEL CMD ATTN: AMCPM-MD-T-G REDSTONE ARSENAL AL 35809

1/85 AFLRL No. 186 Page 2 of 4

CDR US ARMY QUARTERMASTER SCHOOL ATTN: ATSM-CD ATSM-PFS	1	CDR NAVAL SEA SYSTEMS CMD ATTN: CODE 05M4 (MR R LAYNE) WASHINGTON DC 20362	1
FORT LEE VA 23801  HQ, US ARMY ARMOR CENTER AND FORT KNOX ATTN: ATSB-CD FORT KNOX KY 40121	1	CDR DAVID TAYLOR NAVAL SHIP R&D CTR ATTN: CODE 2830 (MR G BOSMAJIAN) CODE 2759 (MR STRUCKO) ANNAPOLIS MD 21402	1 10
CDR US ARMY WESTERN COMMAND ATTN: APLG-TR FORT SCHAFTER HI 96858	1	CG FLEET MARINE FORCE ATLANTIC ATTN: G4 (COL ROMMANTZ) NORFOLK VA 23511	1
CDR US ARMY LOGISTICS CTR ATTN: ATCL-MS (MR A MARSHALL) FORT LEE VA 23801	1	PROJ MGR, M60 TANK DEVELOPMENT ATTN: USMC-LNO US ARMY TANK-AUTOMOTIVE COMMAND (TACOM) WARREN MI 48090	1
CDR US ARMY ENGINEER SCHOOL ATTN: ATZA-CDD FORT BELVOIR VA 22060-5606 CDR	l	DEPARTMENT OF THE NAVY HQ, US MARINE CORPS ATTN: LPP (MAJ WALLER) LMM/3 (MAJ WESTERN) WASHINGTON DC 20380	1
US ARMY INFANTRY SCHOOL ATTN: ATSH-CD-MS-M FORT BENNING GA 31905 CDR	1	CDR NAVAL AIR SYSTEMS CMD ATTN: CODE 53645 (MR MEARNS) WASHINGTON DC 20361	1
US ARMY AVIATION CTR & FT RUCKE ATTN: ATZQ-DI FORT RUCKER AL 36362	ER l	CDR NAVAL RESEARCH LABORATORY ATTN: CODE 6180	1
PROG MGR, TANK SYSTEMS ATTN: AMCPM-MIEI AMCPM-M60 WARREN MI 48090	1	WASHINGTON DC 20375  CDR NAVAL FACILITIES ENGR CTR	
DEPARTMENT OF THE NAVY		ATTN: CODE 1202B (MR R BURRIS) 200 STOVWALL ST ALEXANDRIA VA 22322	1
CDR NAVAL AIR PROPULSION CENTER ATTN: PE-33 (MR D'ORAZIO) P O BOX 7176 TRENTON NJ 06828	1	COMMANDING GENERAL US MARINE CORPS DEVELOPMENT & EDUCATION COMMAND ATTN: DO74 (LTC WOODHEAD) QUANTICO VA 22134	J

1/85 AFLRL No. 186 Page 3 of 4

DR, NAVAL MATERIEL COMMAND		CDR
ATTN: MAT-08E (MR VREATT)	5	USAF 3902 TRANSPORTATION
MAT-O8E (MR ZIEM)	1	SQUADRON
WASHINGTON DC 20360		ATTN: LGTVP (MR VAUGHN) 1
		OFFUTT AIR FORCE BASE NE 68113
CHIEF OF NAVAL OPERATIONS		•
ATTN: OP 413	1	CDR
WASHINGTON DC 20350		HQ 3RD USAF
		ATTN: LGSF (MR PINZOLA)
GG		APO NEW YORK 09127
FLEET MARINE FORCE PACIFIC		
ATTN: G4 (COL HARMS)	1	CDR
CAMP H.M. SMITH HI 96861	_	DET 29
		ATTN: SA-ALC/SFM 1
CDR		CAMERON STATION
NAVY PETROLEUM OFC		ALEXANDRIA VA 22314
ATTN: CODE 43	1	ALEXANDRIA VA 22314
CAMERON STATION	•	
ALEXANDRIA VA 22314		
ALEXANDRIA VA 22314		OTHER GOVERNMENT AGENCIES
DEPARTMENT OF THE AIR FORCE		US DEPARTMENT OF ENERGY
DEPARTMENT OF THE AIR PORCE		CE-1312
		ATTN: MR ECKLUND 1
HQ, USAF	_	FORRESTAL BLDG.
ATTN: LEYSF (COL CUSTER)	1	1000 INDEPENDENCE AVE, SW
WASHINGTON DC 20330		WASHINGTON DC 20585
HQ AIR FORCE SYSTEMS CMD		NATIONAL INSTITUTE FOR PETROLEUM
ATTN: AFSC/DLF (MAJ VONEDA)	1	
	7	AND ENERGY RESEARCH 1
ANDREWS AFB MD 20334	•	AND ENERGY RESEARCH 1 PO BOX 2128
ANDREWS AFB MD 20334	•	
	•	PO BOX 2128
ANDREWS AFB MD 20334		PO BOX 2128
ANDREWS AFB MD 20334 CDR		PO BOX 2128
ANDREWS AFB MD 20334  CDR US AIR FORCE WRIGHT AERONAUTICA	<b>NL</b>	PO BOX 2128
ANDREWS AFB MD 20334  CDR US AIR FORCE WRIGHT AERONAUTICALAB  ATTN: AFWAL/POSF (MR CHURCHILL)	<b>NL</b>	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES)	<b>NL</b>	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS)	AL ) 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES)	AL ) 1 1	PO BOX 2128
ANDREWS AFB MD 20334  CDR US AIR FORCE WRIGHT AERONAUTICA LAB  ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS)  WRIGHT-PATTERSON AFB OH 45433	AL ) 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433 CDR	AL ) 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433 CDR SAN ANTONIO AIR LOGISTICS	AL ) 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433 CDR SAN ANTONIO AIR LOGISTICS CTR	AL 1 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICALAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433 CDR SAN ANTONIO AIR LOGISTICS CTR ATTN: SAALC/SFT (MR MAKRIS)	AL 1 1 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICALAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433 CDR SAN ANTONIO AIR LOGISTICS CTR ATTN: SAALC/SFT (MR MAKRIS) SAALC/MMPRR	AL 1 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICALAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433 CDR SAN ANTONIO AIR LOGISTICS CTR ATTN: SAALC/SFT (MR MAKRIS)	AL 1 1 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433 CDR SAN ANTONIO AIR LOGISTICS CTR ATTN: SAALC/SFT (MR MAKRIS) SAALC/MMPRR KELLY AIR FORCE BASE TX 78241	AL 1 1 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433 CDR SAN ANTONIO AIR LOGISTICS CTR ATTN: SAALC/SFT (MR MAKRIS) SAALC/MMPRR KELLY AIR FORCE BASE TX 78241 CDR	AL 1 1 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433  CDR SAN ANTONIO AIR LOGISTICS CTR ATTN: SAALC/SFT (MR MAKRIS) SAALC/MMPRR KELLY AIR FORCE BASE TX 78241  CDR WARNER ROBINS AIR LOGISTIC	AL 1 1 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433  CDR SAN ANTONIO AIR LOGISTICS CTR ATTN: SAALC/SFT (MR MAKRIS) SAALC/MMPRR KELLY AIR FORCE BASE TX 78241  CDR WARNER ROBINS AIR LOGISTIC CTR	AL 1 1 1 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433  CDR SAN ANTONIO AIR LOGISTICS CTR ATTN: SAALC/SFT (MR MAKRIS) SAALC/MMPRR KELLY AIR FORCE BASE TX 78241  CDR WARNER ROBINS AIR LOGISTIC CTR ATTN: WR-ALC/MMTV (MR GRAHAM)	AL 1 1 1 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433  CDR SAN ANTONIO AIR LOGISTICS CTR ATTN: SAALC/SFT (MR MAKRIS) SAALC/MMPRR KELLY AIR FORCE BASE TX 78241  CDR WARNER ROBINS AIR LOGISTIC CTR	AL 1 1 1 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433  CDR SAN ANTONIO AIR LOGISTICS CTR ATTN: SAALC/SFT (MR MAKRIS) SAALC/MMPRR KELLY AIR FORCE BASE TX 78241  CDR WARNER ROBINS AIR LOGISTIC CTR ATTN: WR-ALC/MMTV (MR GRAHAM)	AL 1 1 1 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433  CDR SAN ANTONIO AIR LOGISTICS CTR ATTN: SAALC/SFT (MR MAKRIS) SAALC/MMPRR KELLY AIR FORCE BASE TX 78241  CDR WARNER ROBINS AIR LOGISTIC CTR ATTN: WR-ALC/MMTV (MR GRAHAM) ROBINS AFB GA 31098	AL 1 1 1 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433  CDR SAN ANTONIO AIR LOGISTICS CTR ATTN: SAALC/SFT (MR MAKRIS) SAALC/MMPRR KELLY AIR FORCE BASE TX 78241  CDR WARNER ROBINS AIR LOGISTIC CTR ATTN: WR-ALC/MMTV (MR GRAHAM) ROBINS AFB GA 31098	AL 1 1 1 1 1	PO BOX 2128
CDR US AIR FORCE WRIGHT AERONAUTICA LAB ATTN: AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) AFWAL/MLSE (MR MORRIS) WRIGHT-PATTERSON AFB OH 45433  CDR SAN ANTONIO AIR LOGISTICS CTR ATTN: SAALC/SFT (MR MAKRIS) SAALC/MMPRR KELLY AIR FORCE BASE TX 78241  CDR WARNER ROBINS AIR LOGISTIC CTR ATTN: WR-ALC/MMTV (MR GRAHAM) ROBINS AFB GA 31098	AL 1 1 1 1 1	PO BOX 2128

CHANGE TO SECURE SECURIOR SECU

SECTION OF STREET, SOUTHERN STREET,